

A decorative graphic consisting of five glowing spheres arranged in a circular pattern. Two spheres are orange and three are cyan. The spheres have a bright center and a soft, glowing outer edge.

# Precise Orbits and Clocks

An introduction to the SP3  
file format used to store  
precise GNSS satellite orbit  
and clock data

# FRONTIER S I >

Rupert Brown  
rbrown@frontiersi.com.au

FrontierSI  
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+61 406 966 992



<https://frontiersi.com.au/>



Door 34, Goods Shed, Village Street, Docklands, VIC, 3008, Australia

# Precise Orbits and Clocks I

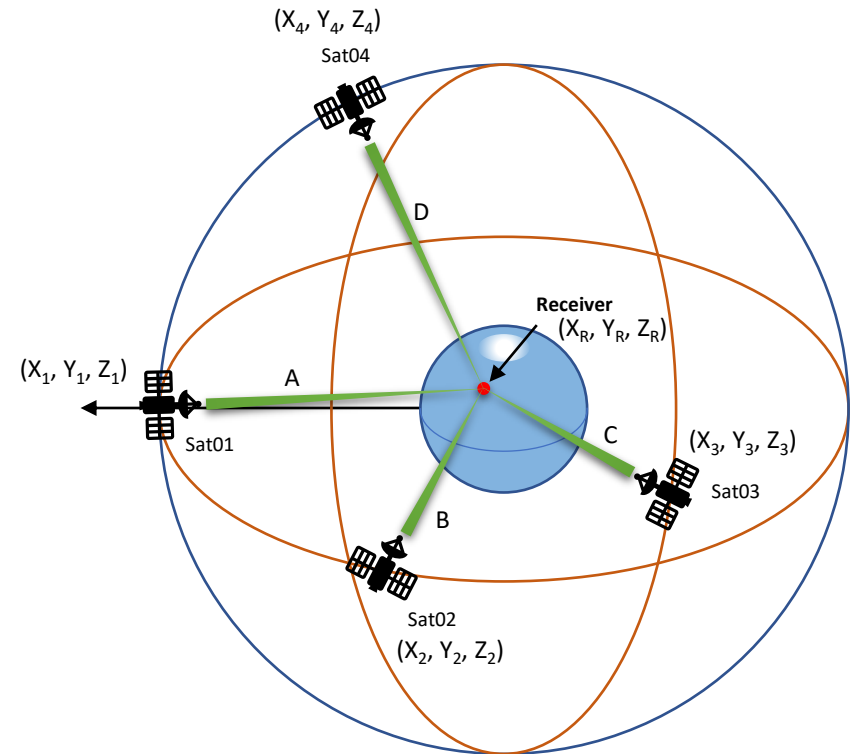
Working out where you are on the Earth using navigation satellites relies on two things:

- Knowing exactly where those satellites are, and
- Knowing exactly how far away you are from each satellite.

If you know those things you can use trigonometry to calculate where you are.

But the nature of global navigation satellite systems means that some errors or unknowns creep in which makes the calculated position less accurate than it could be.

For example, the satellite tells you its position but that position isn't quite right. Its onboard clock is a little out. Both of these things affect your receiver's ability to know how far you are from the satellite and thus reduces the accuracy of the calculated position.



If you know A, B, C and D and all the X, Ys and Zs then you can calculate  $X_R$ ,  $Y_R$  and  $Z_R$ .

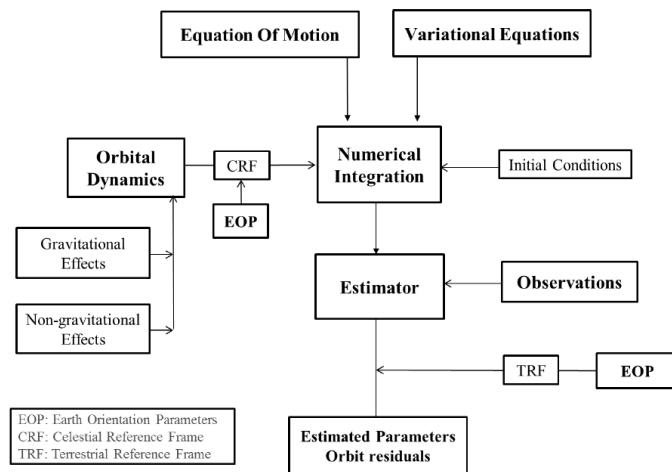
# Precise Orbits and Clocks II

Satellites broadcast data about their orbit (ephemeris) on a regular basis, and this data is used by GNSS receivers to derive positions. The ephemeris data, while accurate, is not as accurate as it could be. The original GPS ephemeris had a precision of about a meter. The newer system Galileo broadcasts data which is three times more precise. As the technology and techniques develop, so will the precision of the satellite broadcast ephemeris data.

Table 3.8: GPS/Galileo/Beidou broadcast ephemeris and clock message parameters.

Parameter	Explanation
$t_{oe}$	Ephemerides reference epoch in seconds within the week
$\sqrt{a}$	Square root of semi-major axis
$e$	Eccentricity
$M_o$	Mean anomaly at reference epoch
$\omega$	Argument of perigee
$i_o$	Inclination at reference epoch
$\Omega_0$	Longitude of ascending node at the beginning of the week
$\Delta n$	Mean motion difference
$\dot{i}$	Rate of inclination angle
$\dot{\Omega}$	Rate of node's right ascension
$c_{uc}, c_{us}$	Latitude argument correction
$c_{rc}, c_{rs}$	Orbital radius correction
$c_{ic}, c_{is}$	Inclination correction
$a_0$	Satellite clock offset
$a_1$	Satellite clock drift
$a_2$	Satellite clock drift rate

A Precise Orbit Determination (POD) system



The clock and orbit data broadcast by the satellites is calculated within the satellite and periodically adjusted by ground control stations. At the same time ground based GNSS reference stations collect GNSS data (observables) and using sophisticated orbit correction models, are able to produce very accurate GNSS satellite orbit and clock data. Precise Point Positioning (PPP) applications, such as Ginan from Geoscience Australia, have their own Precise Orbit Determination (POD) components to do this type of work. These applications usually produce precise orbit and clock data in the SP3-d file format (and as a message in a real-time data stream).

# Precise Orbits and Clocks III

Analysis products such as SP3 files, may be classified according to their latency – the length of time between the time of the last observations and the time the file is produced. The table to the right shows the versions of the orbit and clocks product, differentiated by latency, produced by the International GNSS Service (IGS).

The ultra-rapid product, useful for real-time and near real-time applications, is produced four times per day; the ultra-rapid product includes both observed and predicted satellite orbits. The rapid orbit combination is a daily solution available approximately 17 hours after the end of the previous UTC day. The final, most consistent and highest quality IGS solutions, consist of daily orbit files, generated on a weekly basis approximately 13 days after the end of the solution week.

A crucial role is played by the International GNSS Service (IGS) - up to eight IGS Analysis Centres (AC) contribute daily Ultra-rapid, Rapid and Final GPS orbit and clock solutions to form IGS analysis products. Properly weighted combinations of results from the ACs can produce products which are superior in precision, accuracy, stability, reliability, and robustness compared to the results of any individual AC.

IGS product type, accuracy and latency

Orbit type	3D accuracy		Latency	Updates	
	(metres)	(nsec)			
Broadcast	2.00	7.0	2.10	Real-time	----
Ultra-rapid (predicted)	0.10	5.0	1.50	Real-time	0.25
Ultra-rapid (observed)	<0.05	0.2	0.06	3 hours	0.25
Rapid	<0.05	0.1	0.03	17 hours	1
Final	<0.05	<0.1	<0.03	13 days	7



Figure 1: Weighted orbit RMS of the IGS Rapid (IGR) products and AC Final orbit solutions during 1994-2009 with respect to the IGS Final orbit products.

- COD – Center for Orbit Determination
- EMR – Natural Resources Canada
- ESA – European Space Agency
- GFZ – GeoForschungsZentrum
- JPL – Jet Propulsion Laboratory
- MIT – Massachusetts Institute of Technology
- NGS – National Geodetic Survey
- SIO – Scripps Institute of Oceanography
- IGR – International GNSS Service Rapid products

From: A GUIDE TO USING INTERNATIONAL GNSS SERVICE (IGS) PRODUCTS, Jan Kouba, Geodetic Survey Division, Natural Resources Canada

# SP3-d file overview

Header block – time and configuration data, satellite identification

Data type in file, date and time of file, coordinate system and publishing agency

GPS week and time, epoch intervals

Satellite identification

Satellite orbit accuracy exponents.

Area for other parameters.

Area for comments on the file.

Data block – data on satellites organised by epoch (time)

Items which may be present for each epoch.

Epoch date and time

P record for a specified satellite – X Y Z position coordinates in kilometres,, clock value in microseconds

V record for a specified satellite – X Y Z velocity in decimeters/second, rate of change of clock value in  $10^{-4}$  microseconds/second

EP record for a specified satellite – optional position and clock correlation record. Standard deviation of X Y Z position in mm, standard deviation of the clock correction in picoseconds.

EV record for a specified satellite – optional velocity and clock rate-of-change correlation record. Standard deviation of the X Y Z velocities in  $10^{-4}$  millimetres/second. Standard deviation of the clock correction rate-of-change in units of  $10^{-4}$  picoseconds/second.

# SP3-d file example header block

Note: a real SP-d file does not have blank lines between lines. These have been added below to help fit in the explanatory notes.

Number of epochs (time) contained in file

Time and date of data start:  
Year – month – day of month – hour – minute – second

P: position data  
V: velocity data

File version: SP3-d

GPS week number

Second record identifier

Seconds within GPS week

Number of satellites in the file

Satellite line identifier.  
Must be a minimum of 5 lines – can identify up to 999 satellites.

Satellite orbit accuracy exponent.  
Must be a minimum of 5 lines – matches up with the satellite identifiers above.

File type descriptor:  
G: GPS, M: mixed, R: Glonass,  
L: LEO, S: SBAS, I: IRNSS, E: Galileo,  
C: BeiDou, J: QZSS

Floating-point base number  
used for computing the standard  
deviations for the components of  
the satellite position and velocity.

Comment lines starting with /\*  
There should be a minimum of  
four up to as many are needed.

Type of data used – can be explained in the comment section.

Coordinate system used – typically realisations of ITRF2005 and ITRF2008.

Orbit type: FIT (fitted), EXT (extrapolated), BCT (broadcast), HLM (Helmert)

Agency identifier

Epoch interval in seconds

Fractional part of Julian day start

Modified Julian day start

Satellite identifiers: G – GPS, R – Glonass, E – Galileo, C – Beidou  
I – Indian IRNSS, J – Japanese QZSS, S – SBAS, L – LEO satellites

Os are the fillers to make up the 5 lines

Satellite orbit accuracy exponents.  
For an exponent of 7, the accuracy is  $2^{**}7\text{mm} = \sim 1.3\text{m}$

Os are the fillers to make up the 5 lines

Time system used.  
GPS: GPS, GLO: Glonass, GAL: Galileo, BDT: BeiDou,  
TAI: International Atomic time, UTC: UTC, IRN: IRNSS,  
QZS: QZSS

Floating-point base number for computing the standard  
deviations for the clock correction and the rate-of-change  
of the clock correction.

```

#dP2001 8 8 0 0 0.00000000 192 ORBIT IGS97 HLM MGEX
## 1126 259200.00000000 900.00000000 52129 0.00000000000000
+ 26 G01G02G03G04G05G06G07G08G09G10G11G13G14G17G18G20G21
+ G23G24G25G26G27G28G29G30G31 0 0 0 0 0 0 0 0 0
+ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
+ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
+ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
++ 7 8 7 8 6 7 7 7 7 7 7 7 8 8 7 9
++ 9 8 6 8 7 7 6 7 7 0 0 0 0 0 0 0 0 0 0 0 0
++ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
++ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
++ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
%c G cc GPS ccc cccc cccc cccc cccc cccc cccc cccc cccc
%c cc cc ccc ccc cccc cccc cccc cccc cccc cccc cccc
%f 1.2500000 1.025000000 0.0000000000 0.0000000000000000
%f 0.0000000 0.000000000 0.0000000000 0.0000000000000000
%i 0 0 0 0 0 0 0 0 0
%i 0 0 0 0 0 0 0 0 0
/* AN EXAMPLE ULTRA RAPID ORBIT, GPS ONLY.
/* NOTE THE "PREDICTED DATA" FLAGS FOR THE LAST EPOCH (IN COLUMNS 76 and 80).
  
```

# SP3-d file example - data

Epoch date time

```
* 2001 8 8 0 0 0.00000000
```

Position

Satellite ID	X value	Y value	Z value	Clock
PG01	5783.206741	-18133.044484	-18510.756016	12.734450
PG02	-22412.401440	13712.162332	528.367722	425.364822

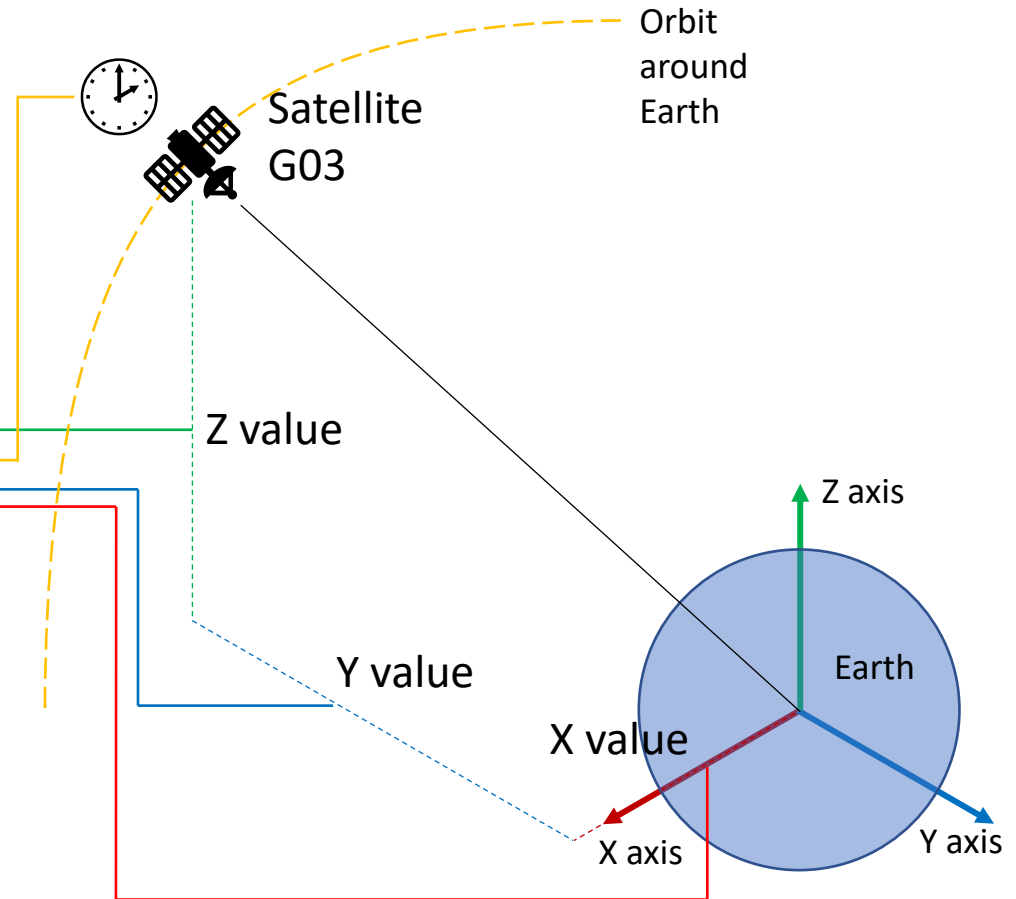
PG03	10114.112309	-17446.189044	16665.051308	189.049475
------	--------------	---------------	--------------	------------

PG04	-24002.325710	4250.313148	-11163.577756	179.333612
PG05	-15087.153141	8034.886396	20331.626539	-390.251167
PG06	13855.140409	-11053.269706	19768.346019	289.556712

Data for another epoch

```
* 2001 8 9 23 45 0.00000000
PG01 4340.761149 -17469.395805 -19521.652181 13.021579
PG02 -22187.015530 13877.264416 2583.141886 425.527461
PG03 9785.535610 -18824.396329 15333.698561 189.465625
PG04 -24642.374460 4816.578416 -9365.337848 180.261632
PG05 -13667.233808 8977.038381 20922.734874 -390.371011
PG06 13696.828033 -12657.020030 18869.219517 288.240920
```

EOF



Note: a real SP-d file does not have blank lines between lines. These have been added below to help fit in the explanatory notes.  
The V, EP and EV record types are not shown in this example.



# SP3-d file example – real file

```
#dP2017 3 29 0 0 0.00000000 96 ORBIT IGS
14 HLM IGS## 1942 259200.00000000 900.00000000 57841 0.00000000000000
+ 32 G01G02G03G04G05G06G07G08G09G10G11G12G13G14G15G16G17
+ G18G19G20G21G22G23G24G25G26G27G28G29G30G31G32 0 0
+ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
+ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
+ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
++ 1 2 2 0 2 2 2 2 2 2 2 2 3 2 2 2 2 2 2
++ 2 2 2 2 2 2 2 2 2 2 1 2 2 2 2 2 0 0
++ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
++ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
++ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
%c G cc GPS ccc cccc cccc cccc ccccc ccccc ccccc ccccc
%c cc cc ccc ccc cccc cccc cccc cccc ccccc ccccc ccccc ccccc
%f 1.2500000 1.025000000 0.00000000000 0.000000000000000
%f 0.0000000 0.000000000 0.00000000000 0.000000000000000
%i 0 0 0 0 0 0 0 0 0 0 0 0
%i 0 0 0 0 0 0 0 0 0 0 0 0
/* FINAL ORBIT COMBINATION FROM WEIGHTED AVERAGE OF:
/* cod emr esa gfz grg jpl mit ngs sio
/* REFERENCED TO IGS TIME (IGST) AND TO WEIGHTED MEAN POLE:
/* PCV:IGS14_1930 OL/AL:FES2004 NONE Y ORB:CMB CLK:CMB
• 2017 3 29 0 0 0.00000000
• PG01 13468.180209 -15714.948137 16388.180018 52.267516 7 4 9 120
• PG02 -14227.390968 -8017.994334 -20506.857548 450.089308 9 11 7 128
• PG03 22329.595919 -13372.180140 -5237.243105 -105.963214 8 7 8 105
• PG04 11936.190982 10697.761976 -21454.649717 999999.999999
• PG05 -23125.831238 -302.341470 -13321.736943 -53.443172 10 6 8 100
• PG06 -7413.463000 -21059.824533 -14352.466041 330.665316 8 9 7 92
• PG07 7221.007672 -24775.126497 -5078.259847 368.285870 9 8 11 86
• PG08 20725.303414 1016.712981 16687.809065 -54.389560 6 7 8 114
• PG09 1990.137990 -16694.765350 -20586.132180 331.550108 7 7 5 102
• PG10 5046.792791 14399.425428 21761.443567 -86.596028 4 4 6 108
• PG11 13521.926322 -10290.651944 19860.647962 -677.801326 5 6 7 113
```

## SP File Format – using

An SP3 file contains the precise position and clock of a satellite at a certain point in time (epoch).

No position or clock is absolutely correct and the file contains data which gives the user an indication of the accuracy of the data.

This in part is driven by the type of file – final, rapid or ultra-rapid. The final files are the most accurate, but the ultra-rapid (and stream data) are obviously most useful for near real-time work.

Positions are given at certain points in time. As a user, if you need to determine a position at a time between two given points, you will have to use an orbital model and interpolate between the two known points to derive a position at the time required. Orbits are predictable so this is a very practical and useful thing to do.

Clocks, due to their noise characteristics are more problematic. The closer the two points are in time the better.

## SP File Format – more information

SP stands for Standard Product and the first file specification SP1 was first released in 1985 by Benjamin Remondi working for the US National Geodetic Survey. At the time the only satellite based navigation constellation in operation was the US Global Positioning System (GPS) and SP1 was focussed on being a means of distributing precise GPS orbit data.

Since 1985 the file specification has undergone many revisions to enhance the data it can contain. In 2016 the SP3-d specification was released. This format supports:

- Satellite identification numbers from all the global navigation satellite systems (GNSS) and satellite based augmentation systems (SBAS) currently in operation, and up to 999 individual satellites,
- Precise orbit details but also clock corrections,
- Clock event and orbit manoeuvre flags,
- More generous space for comments.

For more information on the SP3-d format please refer to [1]. For more information on the history of the SP format please refer to [1] and [2].

[1] The Extended Standard Product 3 Orbit Format (SP3-d), 21 February 2016, Steve Hilla, National Geodetic Survey, National Ocean Service, NOAA, Silver Spring, MD 20910, USA.

[2] NOAA Technical Report NOS 133 NGS 46, Extending the National Geodetic Survey Standard GPS Orbit Formats, Benjamin W. Remondi, Rockville, MD, November 1989

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